

(( CH - 2 ))

(( Fluid Statics ))

In static fluids there is no relative motion ( $\frac{dv}{dy} = 0$ ) ~~thus~~ between fluid elements, thus no shear stress can exist, viscosity has no effects.

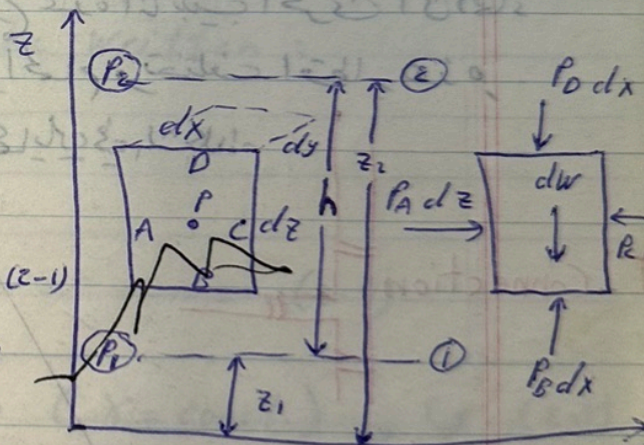
The basic equation of fluid statics is that relating pressure, density & vertical distance

Pressure - Density - Height Relationships :-

Applying Newton's first law to the element ( $dx dz$ ) and using average pressures on each face :-

$$\sum F_x = 0 \Rightarrow P_A \cdot dz - P_C \cdot dz = 0 \quad (2.1)$$

$$\sum F_z = 0 \Rightarrow P_B \cdot dx - P_D \cdot dx - dw = 0 \quad (2.2)$$



Which  $P$  and  $\rho$  are functions of  $x$  and  $z$ . Writing  $P_A$ ,  $P_B$ ,  $P_C$ , and  $P_D$  in terms of pressure ( $P$ ) at the center

$$P_A = P - \frac{\partial P}{\partial x} \frac{dx}{2}, \quad P_C = P + \frac{\partial P}{\partial x} \frac{dx}{2}$$

$$P_B = P - \frac{\partial P}{\partial z} \frac{dz}{2}, \quad P_D = P + \frac{\partial P}{\partial z} \frac{dz}{2}$$

$$dw = \rho dx dz \quad \left\{ \text{where } \rho = \text{const. as } dx \neq dz \rightarrow 0 \right\}$$

Thus, equation (2.1) and (2.2) become:

z

$$\left[ p - \frac{\partial p}{\partial x} \frac{dx}{2} \right] dz - \left[ p + \frac{\partial p}{\partial x} \frac{dx}{2} \right] dz = 0$$

$$-\frac{\partial p}{\partial x} dx \cdot dz = 0 \Rightarrow \boxed{\frac{\partial p}{\partial x} = 0} \quad (2.2)$$

((بالنسبة للمختار يكون الضغط ثابتاً))

$$\left[ p - \frac{\partial p}{\partial z} \frac{dz}{2} \right] dx - \left[ p + \frac{\partial p}{\partial z} \frac{dz}{2} \right] dx - \gamma dx dz = 0$$

$$-\frac{\partial p}{\partial z} dz \cdot dx - \gamma dx dz = 0$$

$$\boxed{\frac{\partial p}{\partial z} = \frac{dp}{dz} = -\gamma} \quad (2.3)$$

((بالنسبة ل z يكون الضغط متغيراً في نفس الضغط كلما ازداد الارتفاع z))

The second of Eq (2.3) may be written:

$$\int_{z_2}^{z_1} -dz = \int_{p_2}^{p_1} \frac{dp}{\gamma} \Rightarrow$$
$$z_2 - z_1 = \int_{p_2}^{p_1} \frac{dp}{\gamma} \quad (2.4)$$

Incompressible fluid :- (سائل غير قابل للانضغاط)

For fluids of constant density ( $\gamma = \text{const.}$ ), Eq. (2.4) becomes:

$$\boxed{p_1 - p_2 = \gamma (z_2 - z_1) = \gamma h} \quad (2.5)$$

$$\text{or } \frac{p_1}{\gamma} + z_1 = \frac{p_2}{\gamma} + z_2 = \text{const.} \quad (2.6)$$

Ex<sup>①</sup> :- The liquid oxygen (LOX) tank of a Saturn moon rocket is partially filled to a depth of 10 m with (LOX) @  $-126^\circ\text{C}$ . The absolute pressure in the vapor above the liquid surface is maintained @ 101.3 kPa calculate the

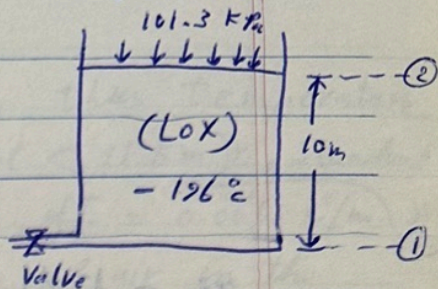
the pressure @ the inlet valve @ the bottom of the tank ?  $\rho = 1206 \text{ kg/m}^3$  (density)

$$P_2 - P_1 = \rho (z_1 - z_2) = \rho h$$

$$P_1 = P_2 + \rho h$$

$$\rho_{\text{Lox}} = \rho g = 9.81 (1206) = 11.8 \text{ kN/m}^3$$

$$P_1 = 101.3 + 10(11.8) = 219.3 \text{ kPa}$$



\* For fluids of variable density ( $\rho \neq \text{const.}$ ) Eq. (2-4) cannot be integrated unless relation between  $\rho$  &  $z$  is known. For gases the polytropic equation:

$$\frac{P}{\rho^n} = \text{const.} \quad (2-7)$$

can be used [For adiabatic  $n = k$ , for isothermal  $n = 1$ ]. Since temperature lapse rate (معدل التغير في درجة الحرارة)  $(-dT/dz)$  is important in atmosphere, substitute  $(\rho RT/g)$  for  $P$  in Eq. (2-7) :-

$$\frac{\rho RT}{g \rho^n} = \text{const.}$$

So, temperature lapse rate in atmosphere :-

$$\boxed{-\frac{dT}{dz} = \frac{g}{R} \left( \frac{n-1}{n} \right)} \quad (2-8)$$

For isothermal atmosphere  $\Rightarrow n = 1 \Rightarrow -\frac{dT}{dz} = 0$

For adiabatic  $\Rightarrow n = k$  (for air  $k = 1.4$ )

$$-\frac{dT}{dz} = \frac{9.81}{286.8} \left( \frac{1.4-1}{1.4} \right) = 0.0028 \text{ } ^\circ\text{C/m} = 9.8 \text{ } ^\circ\text{C/km}$$

In atmosphere  $n > 1$  and  $-dT/dz > 0$ , thus temperature declines with altitude. ( $n = 1.235$  for level  $< 11 \text{ km}$ ), Standard atmosphere ( $\text{@ sea level } T = 15^\circ\text{C}, P = 101.3 \text{ kPa}, -\frac{dT}{dz} = 0.0065 \text{ } ^\circ\text{C/m}$ )

Ex 2:- Calculate pressure and weight density of air in the U.S. standard atmosphere @ altitude 10 km? ~~Calculate pressure and weight density of air in the U.S. standard atmosphere @ altitude 10 km?~~

from table  $\Rightarrow -\frac{dT}{dz} = 0.0065, n = 1.235$   
 $P_1 = 101.3 \text{ kPa}, T_1 = 15^\circ\text{C}, \gamma_1 = 12.01 \text{ N/m}^3$

$$\frac{P_1}{\gamma_1^n} = \frac{P_2}{\gamma_2^n} \Rightarrow \frac{P_1}{\gamma_1^{1.235}} = \frac{101300}{(12.01)^{1.235}}$$

$$\frac{1}{\gamma_1} = \left( \frac{4702.99}{P_1} \right)^{1/1.235} = \frac{941}{P_1^{1/1.235}}$$

$$\frac{dP}{dz} = -\gamma \Rightarrow -dz = \frac{dP}{\gamma} = \frac{941 dP}{P^{1/1.235}}$$

$$-\int_{z_1}^{z_2} dz = \int_{P_1}^{P_2} \frac{941 dP}{P^{0.81}}$$

$$z_1 - z_2 = 0 - 10000 = \frac{941 P^{0.19}}{0.19} \Bigg|_{101.3}^{P_2}$$

$$\Rightarrow P_2 = 26.3 \text{ kPa}$$

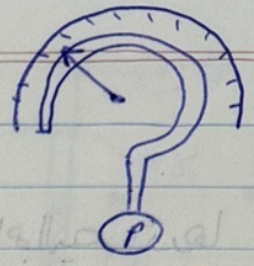
$$-\frac{dT}{dz} = 0.0065 \text{ } ^\circ\text{C/m}$$

$$T_2 = T_1 + dT \Rightarrow T_2 = 15^\circ\text{C} - 0.0065 * 10 * 10^4 = -50^\circ\text{C} = 223 \text{ K}$$

$$\frac{P_1}{\gamma_1^{1.235}} = \frac{P_2}{\gamma_2^{1.235}} \Rightarrow \gamma_2 = \gamma_1 \left[ \frac{P_2}{P_1} \right]^{0.18} = 4.03 \text{ N/m}^3$$

Absolute and Gage Pressures :-

$$\text{Absolute pressure} = \text{atmospheric pressure} - \text{Vacuum} + \text{Gage pressure}$$

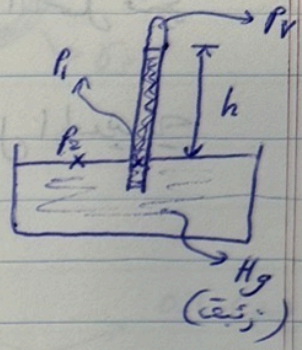


Bourdon gage

$$P_1 = P_2$$

$$P_1 = P_v + h \gamma_{Hg}$$

$$P_v + h \gamma_{Hg} = P_2 = P_{atmosphere}$$



$$-0.17 Pa + h * 13.57 * 9800 = 0$$

$$h = 0.0012 \text{ mm Hg}$$

Ex :- A Bourdon gage registers a vacuum of 310 mm of mercury when the atmospheric (absolute) pressure is 100 kPa. Calculate the corresponding absolute pressure?

$$P_{vac} = 0.31 * 13.57 * 9800 = 41.3 \text{ kPa}$$

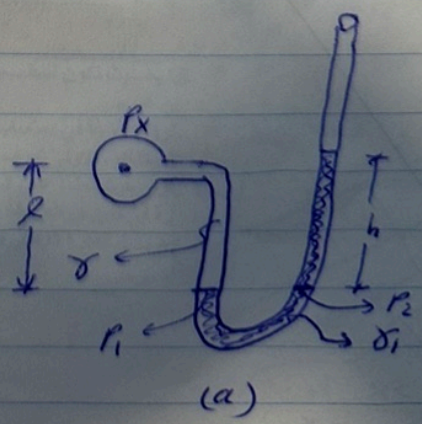
$$P_{absolute} = 100 - 41.3 = 58.7 \text{ kPa}$$

Manometry :-

For fig. (a) :-

$$P_x + l \gamma = h \gamma_1$$

$$P_x = h \gamma_1 - l \gamma$$



(a)

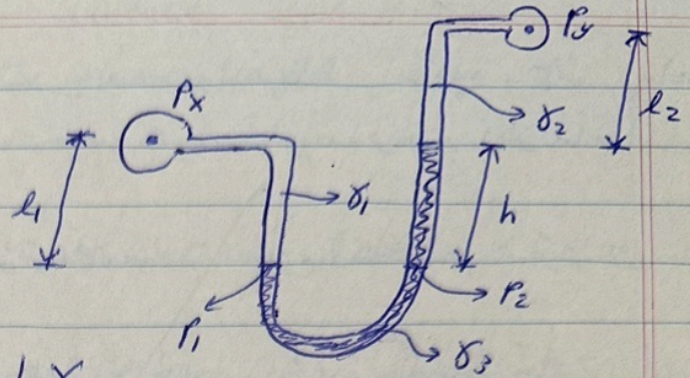
$$P_x + \gamma_1 l_1 = P_1$$

$$P_y + \gamma_2 l_2 + h \gamma_3 = P_2$$

$$P_1 = P_2$$

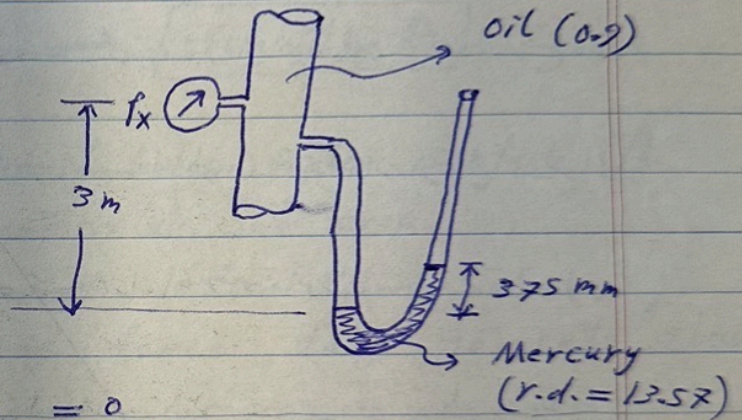
$$P_x + \gamma_1 l_1 = P_y + \gamma_2 l_2 + h \gamma_3$$

$$(P_x - P_y) = \gamma_2 l_2 + h \gamma_3 - \gamma_1 l_1$$



EX:

What will be the gage reading ( $P_x$ )?



$$P_x + (0.9 * 9.8 * 10^3) * 3 - 0.375 * 13.57 * 9800 = 0$$

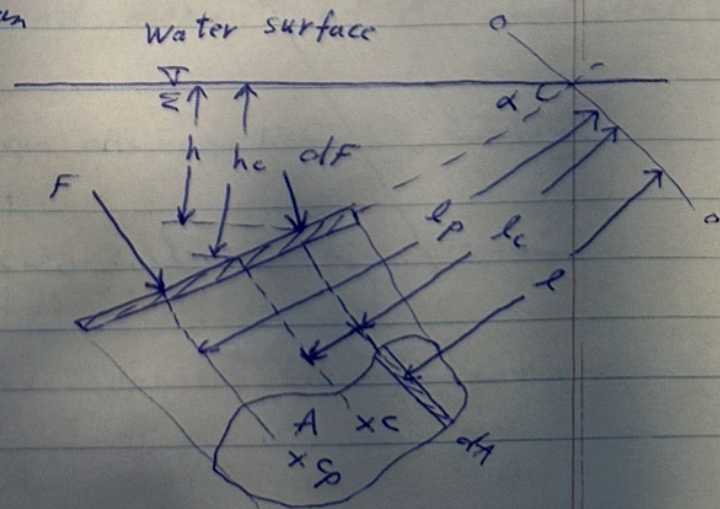
$$P_x = 23.4 \text{ KPa}$$

Forces on Submerged Plane Surfaces : - القوة على السطح المسطح المغمور

The plane area is inclined on angle ( $\alpha$ ) to the water surface. The force on a differential element ( $dA$ ) is:

$$dF = P dA$$

(Note that  $dA$  is taken so is constant over it)



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$h =$  depth below the water surface  $= l \sin \alpha$

$l =$  distance from 0-0 to element  $dA$  along the plane area.

$$dF = \gamma l \sin \alpha dA$$

$$F = \int_A \gamma l \sin \alpha dA = \gamma \sin \alpha \int_A l dA$$

Where  $\int_A l dA =$  first moment of area ( $A$ ) about 0-0 and equal to  $(l_c \cdot A)$ , where  $l_c$  is the distance of centroid of  $A$  from 0-0.

$$\therefore F = \gamma \sin \alpha l_c A \Rightarrow \boxed{F = \gamma h_c A}$$

$h_c =$  the depth of centroid below water surface. The direction of the force is perpendicular to area.

To find the location of force, take moments about 0-0 :

$$dM = l dF = l \gamma l \sin \alpha dA = \gamma l^2 \sin \alpha dA$$

$$M = \gamma \sin \alpha \int_A l^2 dA$$

Where  $\int_A l^2 dA$  is the second moment of  $A$  about 0-0.

$$\boxed{M = \gamma \sin \alpha I_{0-0}}$$

The location of force is :-

$$l_p = \frac{M}{F} = \frac{\gamma \sin \alpha I_{0-0}}{\gamma h_c A} = \frac{\gamma \sin \alpha I_{0-0}}{\gamma l_c \sin \alpha A}$$

$$= \frac{I_{0-0}}{l_c \cdot A} = \frac{I_c + l_c^2 A}{l_c A} = l_c + \frac{I_c}{l_c A}$$

$$l_p - l_c = \frac{I_c}{l_c A}$$

Where  $l_p - l_c$  = distance between centroid of area and center of pressure (Point of application of force).

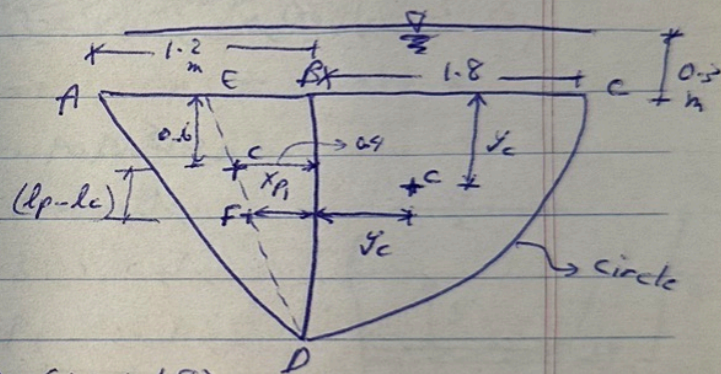
Ex. - Calculate magnitude, direction, and location of the total force exerted by the water on one side of this composite area which lies in a vertical plane?

From appendix (6)

for quarter circle

$$y_c = \frac{4r}{3\pi} = \frac{4 * 1.8}{3\pi} = 0.764 \text{ m}$$

$$F = \gamma h_c A$$



$$F)_{\text{triangle}} = 9800 (0.3 + 0.6) \left( \frac{1.2 * 1.8}{2} \right) = 9.53 \text{ kN}$$

$$F)_{\text{circle}} = 9800 (0.3 + 0.764) \frac{\pi (1.8)^2}{4} = 26.53 \text{ kN}$$

$$\text{Total force} = 9.53 + 26.53 = 36.06 \text{ kN}$$

To find vertical location of force :-

$$\begin{aligned} \text{For triangle } (l_p - l_c) &= \frac{I_c}{l_c A} = \frac{bh^3/36}{(0.3 + 0.6) \frac{bh}{2}} \\ &= \frac{1.2 * 1.8^3/36}{(0.9) \frac{1.2 * 1.8}{2}} = 0.2 \text{ m} \end{aligned}$$

2

$$(I) \text{ for semi-circle} = \frac{\pi d^4}{128}$$

$$\therefore (I) \text{ for quadrant} = \frac{\pi d^4}{256} = \frac{\pi (2 \times 1.8)^4}{256} = 2.061 \text{ m}^4$$

$$I = I_c + y_c^2 A \Rightarrow I_c = I - y_c^2 A$$

$$= 2.061 - (0.764)^2 (2.54)$$

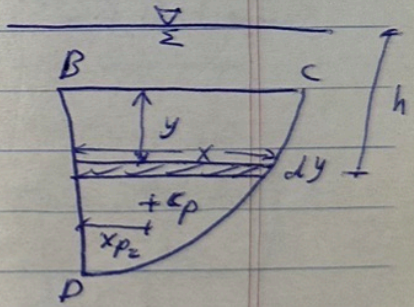
$$= 0.576 \text{ m}^4$$

$$l_p - l_c = \frac{I_c}{l_c A} = \frac{0.576}{1.064 \times 2.54} = 0.213 \text{ m}$$

~~Handwritten scribbles and crossed-out text.~~

$$\frac{0.4 - x_{p1}}{0.2} = \frac{0.6}{1.8} \Rightarrow x_{p1} = 0.333 \text{ m}$$

For location of force on quadrant  
take moment about BD :-



$$dM = \frac{x}{2} dF$$

$$= \frac{x}{2} \delta h dA$$

$$= \frac{x}{2} \delta h x dy$$

$$\left\{ h = 0.3 + y, \quad x^2 + y^2 = (1.8)^2 \right\}$$

$$\int dM = \int \delta \frac{x^2}{2} (0.3 + y) dy$$

$$M = \frac{9800}{2} \int_{0.3}^{1.8} (1.8^2 - y^2) (y + 0.3) dy$$

$$= 18575 \text{ N.m}$$

$$x_{p2} = \frac{M}{F} = \frac{18575}{26.53 \times 10^3} = 0.7 \text{ m}$$

$$36.06 X_p = 26.53 * 0.7 - 9.53 * 0.333$$

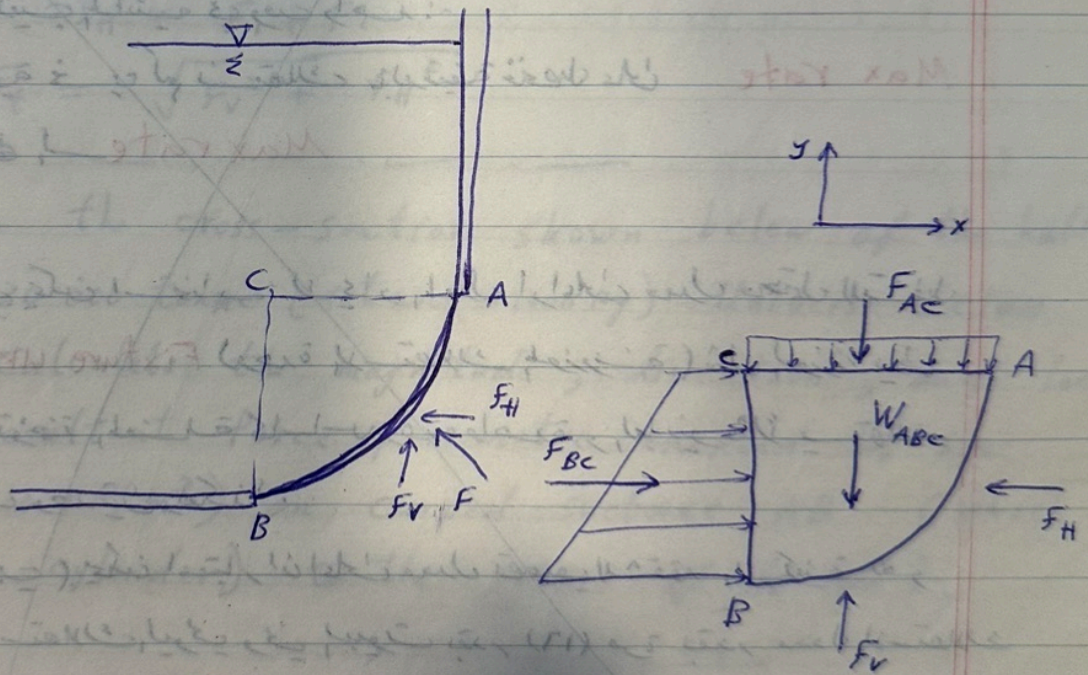
$$X_p = 0.427$$

take moment about (Ac) for vertical location:

$$36.06 \bar{y} = 26.53 (0.764 + 0.213) + 9.53 (0.3)$$

$$\bar{y} = 0.93 \text{ m}$$

Forces on Submerged Curved Surface:



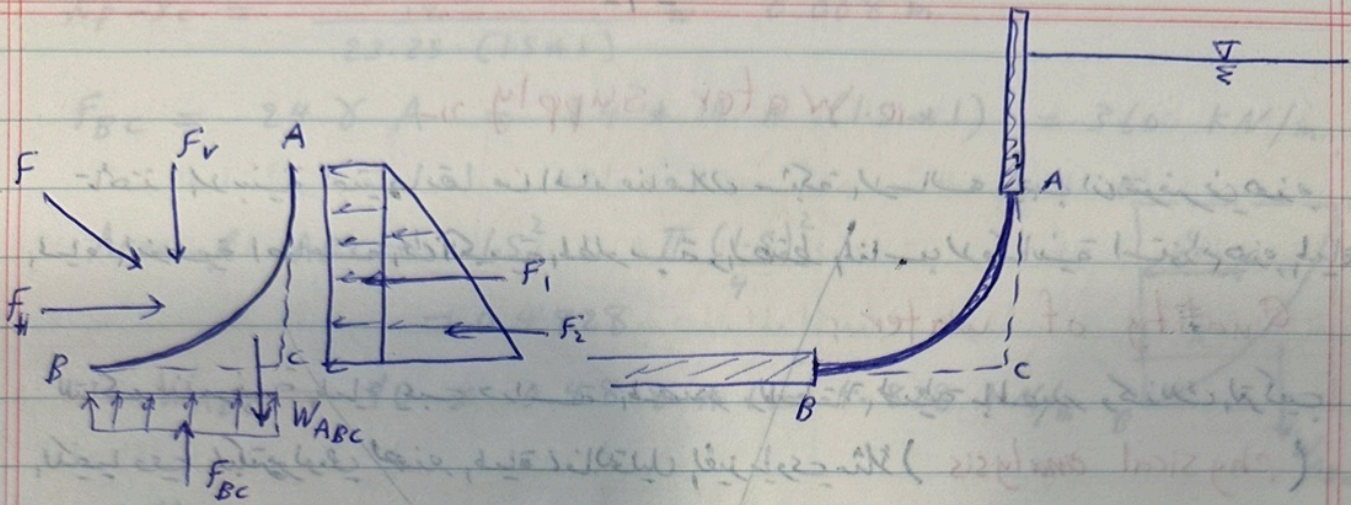
$$\sum \vec{F}_x = 0$$

$$F_H = F_{BC}$$

$$\downarrow \sum F_y = 0$$

$$F_{AC} + W_{ABC} = F_V$$

$$F = \sqrt{F_H^2 + F_V^2}$$



$$F_v = F_{BC} - W_{ABC}$$

$$F_H = F_1 + F_2$$

$$F_t = \sqrt{F_v^2 + F_H^2}$$

Ex: Consider the cross-section shown below of the hull of typical 330 000 tonne (1 tonne = 1000 kg) Universe class oil tanker. Calculate the magnitude, direction, and location of the resultant force per metre exerted by sea water ( $\gamma = 10 \text{ kN/m}^3$ ) on curved surface AB (which is a quarter cylinder) @ the corner of the hull?

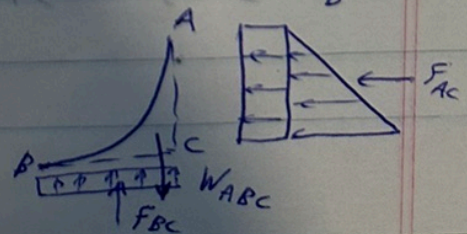
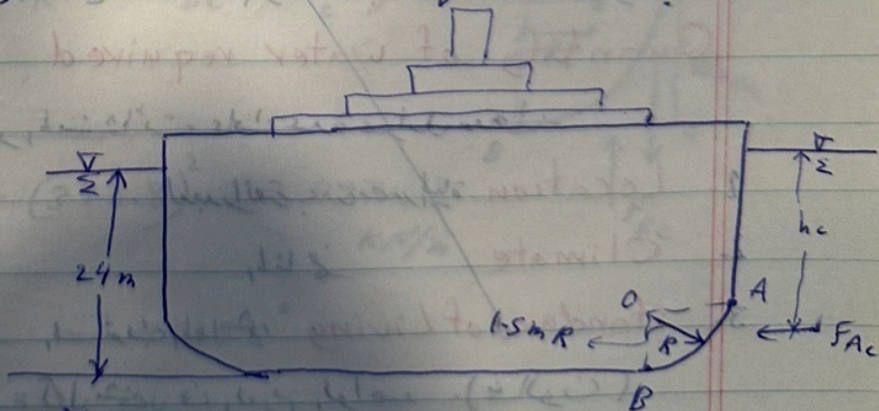
$$F_{Ac} = \gamma h_c A$$

$$= 10 * (24 - \frac{1.5}{2})$$

$$(1.5 * 1)$$

$$= 348.8 \text{ kN/m} = F_H$$

$$l_p - l_c = \frac{I_c}{l_c A} = \frac{bh^3/12}{l_c \cdot A}$$

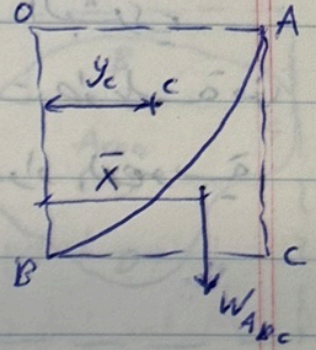


$$I_p - I_c = \frac{1.5^3 * 1}{12} - 23.25 (1.5 * 1) = 0.008 \text{ m}$$

$$F_{BC} = 24 \delta A = 24 * 10 * (1.5 * 1) = 360 \text{ kN/m}$$

$$\text{Area of } ABC = 1.5^2 - \frac{\pi (1.5)^2}{4} = 0.4828$$

$$W_{ABC} = 10 * 0.4828 * 1 = 4.828 \text{ kN/m}$$



$$\sum M_{O-B} = 0$$

$$\frac{\pi r^2}{4} y_c + A_{ABC} * \bar{X} = 1.5^2 \left(\frac{1.5}{2}\right)$$

$$\frac{\pi (1.5)^2}{4} \left(\frac{3 * 1.5}{4 \pi}\right) + 0.4828 \bar{X} = 1.5^2 \left(\frac{1.5}{2}\right)$$

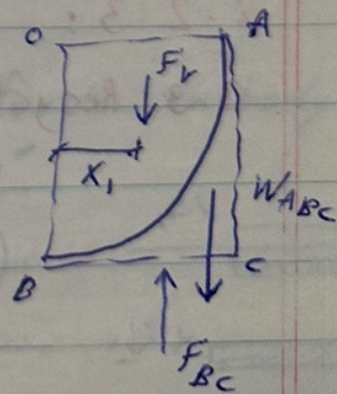
$$\bar{X} = 1.165 \text{ m to the right of } B-O$$

$$F_v = F_{BC} - W_{ABC} = 355.2 \text{ kN/m} \downarrow$$

Location of Vertical Force :-

$$F_v * x_1 + W_{ABC} * \bar{X} = F_{BC} \left(\frac{1.5}{2}\right)$$

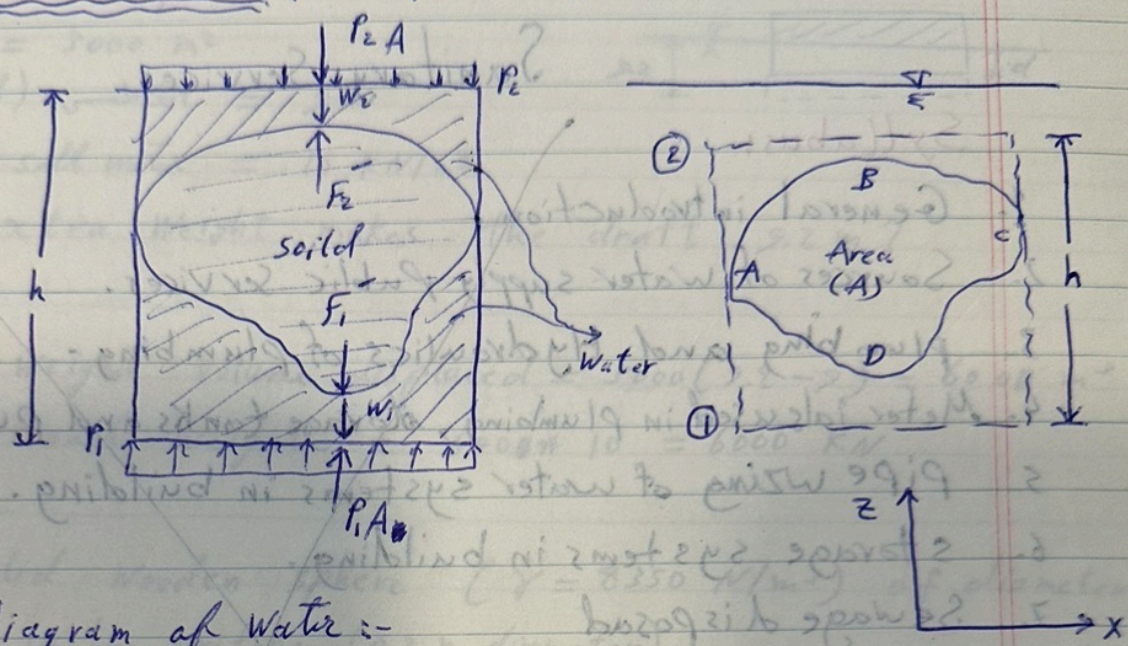
$$x_1 = 0.74 \text{ m}$$



$$F_R = \sqrt{(355.2)^2 + (348.8)^2} = 497.8 \text{ kN/m}$$

$$\theta = \tan^{-1} \frac{F_v}{F_H} = 45.52^\circ$$

Buoyancy and Flotation : الطفو والعموم



Free body diagram of water :-

For upper portion

$$\uparrow \Sigma F_z = 0 \Rightarrow F_2' - P_2 A - W_2 = 0$$

Lower portion :-

$$\uparrow \Sigma F_z = 0 \Rightarrow P_1 A - F_1' - W_1 = 0$$

$$F_2' = W_2 + P_2 A$$

$$F_1' = P_1 A - W_1$$

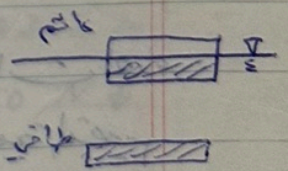
$F_B = \text{Buoyancy force} = \text{net upward force on body}$

$$= \text{Force up} - \text{Force down} = F_1' - F_2'$$

$$F_B = F_1' - F_2' = (P_1 A - W_1) - (P_2 A + W_2)$$

$$= (P_1 - P_2) A - (W_1 + W_2)$$

$$= \delta h A - (W_1 + W_2)$$



The right hand side is the weight of the volume of liquid exactly equal to volume of the body.

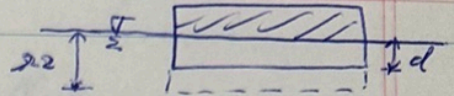
$$F_B = \delta (\text{Volume of object})$$

$$F_B = \delta (\text{Volume displaced})$$

$$F_B = \text{Weight of object}$$

Ex:- A container ship :-

Area =  $3000 \text{ m}^2$   
 draft (d) =  $9 \text{ m}$



$\gamma$  for salt water =  $10 \text{ kN/m}^3$

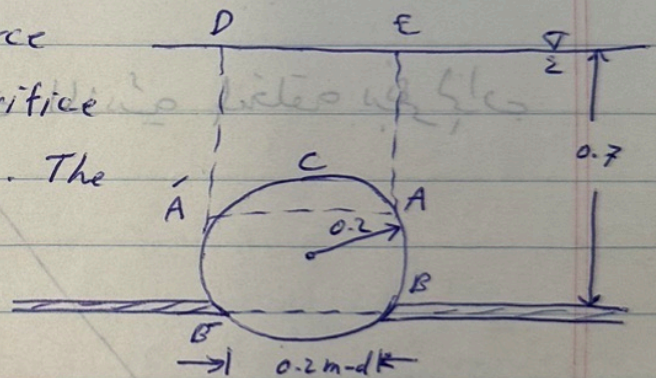
How much extra weight makes the draft  $9.2 \text{ m}$ ?

Additional weight volume displaced =  $3000(9.2 - 9) = 6000 \text{ m}^3$

Additional weight =  $6000 * 10 = 6000 \text{ kN}$

Ex:- The Solid wooden sphere ( $\gamma = 8350 \text{ N/m}^3$ ) of diameter  $0.4 \text{ m}$  is held in the orifice ( $0.2 \text{ m}$  diameter)

by the water. Calculate the force exerted between sphere and orifice plate when the depth is  $0.7 \text{ m}$ . The sphere will float away if this force becomes zero; can this ever happen?



Volume of sphere =  $\frac{4\pi}{3} R^3 = \frac{4\pi}{3} (0.2)^3 = 0.0335 \text{ m}^3$

Weight of sphere =  $0.0335 * 8350 = 280 \text{ N} \downarrow$

Volume of sphere outside cylinder (BACA') =  $0.0218 \text{ m}^3$  (from geometry)

$F_B = 0.0218 * 9800 = 214 \text{ N} \uparrow$

Volume of (A'CAED) =  $0.0098 \text{ m}^3$

Force of (A'CAED) =  $9800 * 0.0098 = 96 \text{ N} \downarrow$

∴ Net Force (Force between sphere and plate)

$$= 280 + 96 - 214 = 162 \text{ N} \downarrow$$

$$\therefore F_{\text{net}} = 280 - 214 = 66 \text{ N}$$

∴ The force between sphere and plate can never be zero.